

Module 3: Mechanical Properties of Materials and Application

Syllabus:

Elastic Moduli, Poisson's ratio, Relation between Y , n and σ (with derivation), mention relation between K , Y and σ , limiting values of Poisson's ratio, Factors affecting the elastic properties. Torsion of a cylinder: Qualitative discussion on Torsion of cylinder and mention the expression for a couple per unit twist of a solid cylinder, Cantilever (derivation of expression with fundamentals of bending of beams), Numerical Problems

Introduction

The study of strength of materials is to provide the means of analysing and designing various machines and load bearing structures. Elasticity is an elegant and fascinating subject that deals with determination of the stress, strain and displacement distribution in an elastic solid under the influence of external forces. Following the assumptions of linear, small deformation theory, the formulation establishes a mathematical model providing solutions to problems that have applications in many engineering and scientific fields. Civil engineering applications include stress and deflection analysis of structures like rods, beams, plates, shells soil, rock concrete etc. Mechanical engineering uses elasticity in numerous problems of thermal stress analysis, fracture mechanics, fatigue, and design of machine elements. Material engineering uses elasticity to determine the stress fields of crystalline solids, dislocations, microstructures etc. Applications in aeronautical engineering include stress fluctuations, fracture, fatigue analysis in aero structures. The subject also provides the basis for study of materials behaviour in plasticity and viscoelasticity.

Elasticity:

*“The property of material body to regain its original shape and size on removal of the deforming forces is called **elasticity**”.*

Within certain elastic limit steel and quartz show elastic properties. The elastic property is desirable for materials used in tools and machines.

Importance of Elasticity in Engineering Applications:

A sound knowledge of elastic properties of materials is very essential in the field of engineering. Engineers can make better choice of the materials for their use by knowing the nature of its stress-strain curve. Pure metals are soft by property. They are ductile and have low tensile strength. Hence, they are rarely used in engineering applications. Alloys are generally harder than pure metals. They are produced by blending (mixing) different metals after which, they exhibit unique properties that are different from metals mixed. These alloys offer better elastic properties useful for engineering applications.

Plasticity:

A body which does not show any tendency to recover their original condition are said to be Plastic and the property is called plasticity.

E.g. – polyethylene, Polystyrene etc.

This property of the material is necessary for forging, stamping images on coins and ornamental works.

Load:

The term load implies the combination of external forces acting on a body and its effect is to change the form or the dimensions of the body. It is essentially a deforming force.

Stress:

When the deforming forces acting on a body, the restoring or recovering force per unit area of cross section set up inside the body is called stress.

$$\text{Stress} = \frac{\text{Restoring Force}}{\text{cross sectional area}} = \frac{F}{A}, \text{ S I unit of stress is N/m}^2.$$

Types of stress:

1. Tensile Stress:

When a body or section is subjected to two equal and opposite pulls and if it tends to pull apart the particles of the material causing extension in the direction of application of load, then the load is called **tensile load** and the corresponding stress induced is known as **tensile stress**.

Longitudinal stress or tensile stress is applied along the length and hence causes change in length.

2. Compressive stress (Pressure or volume stress):

When a body or section is subjected to two equal and opposite pushes and it tends to push the particles of the material nearer causing shortening in the direction of application of load, then the load is called **compressive load** and the corresponding stress induced is known as **compressive stress**.

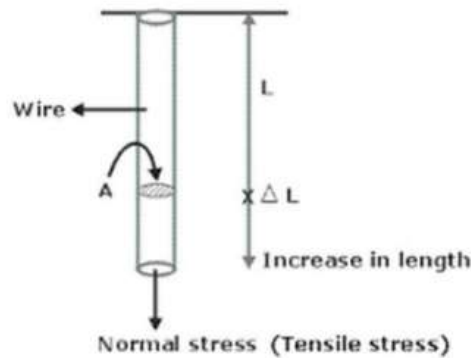
3. Tangential stress or Shearing stress:

When a body is subjected to a force acting along the tangential direction, the body experiences a turning or twisting effect resulting in the change in the shape of the body without any change in its volume, the stress induced is called **shearing stress** or **tangential stress**.

Strain:

When a body is subjected to external force, there will be change in dimensions of the body. The change in dimension is called deformation.

“The ratio of change in dimension of body or deformation to the original dimension of the body is called known as strain”.



If a bar is subjected to a direct load and hence a stress, the bar will change in length. If the bar has an original length L and changes by an amount ΔL , the strain produced is defined as follows:

$$\text{Strain} = \frac{\text{change in dimension of the body}}{\text{original dimension}} = \frac{\Delta L}{L}$$

Strain is thus, a measure of the deformation of the material and is a non-dimensional quantity i.e., it has no units.

Since in practice, the extensions of materials under load are very very small, it is often convenient to measure the strain in the form of strain $\times 10^{-6}$ i.e., micro strain.

Types of Strain:

1. Linear or Longitudinal Strain:

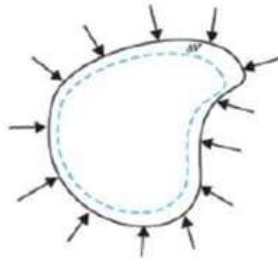
When a force is applied, the ratio of change in the length to the original length of a body is known as longitudinal strain.

$$\text{Longitudinal Strain} = \frac{\text{change in length of the body}}{\text{original length}} = \frac{\Delta L}{L}$$

2. Volume strain:

When uniform pressure is applied normally on all over the surface of a body, the body undergoes a change in its volume. The ratio of change in the volume to the original volume is called as volume strain.

$$\text{Volume strain} = \frac{\Delta V}{V}$$



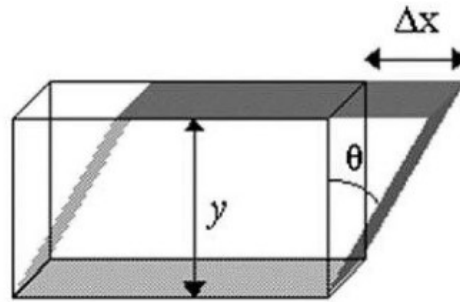
3. Shear Strain:

When a body is subjected to tangential force, the angular displacement of a reference line in the body is as known shear strain. The shearing angle itself is the measure of the ratio of change in dimension to the original dimension.

$$\text{Shearing strain} = \tan\theta = \Delta x / y$$

When the angle of shear is small,

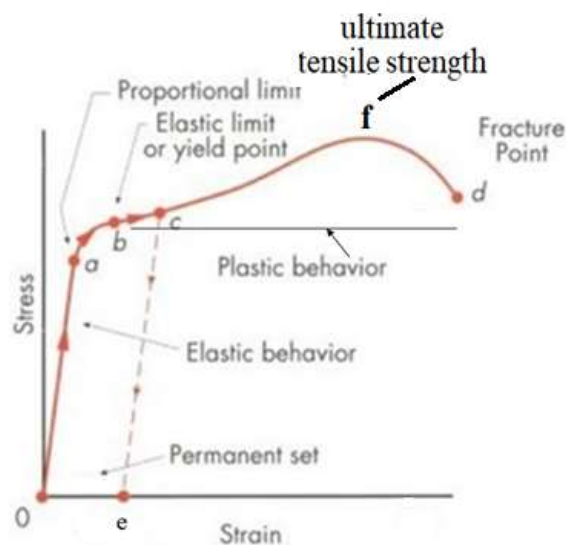
$$\text{Shearing strain} = \theta = \Delta x / y$$



Stress-strain curve of metallic wire:

The relationship between stress and strain is studied by plotting a graph for various values of stress and the accompanying strain and is known as stress-strain diagram.

1. The linear part **oa** of the curve shows that the strain produced is directly proportional to the stress or the Hooke's law is obeyed perfectly up to a. The stress corresponding to the point '**a**' is known as proportional limit beyond which Hooke's law is not obeyed.
2. With an increase in the stress beyond '**a**' the strain increases more rapidly and the curve has smaller and smaller slope until the point '**b**'. Up to '**b**' material exhibits perfect elasticity but does not Hooke's law. The stress at which the linear relationship between stress and strain ceases to hold good is referred to as the **elastic limit or yield strength** of the material.



The stress strain graph for a material is as shown in the fig.

3. As soon as the elastic limit is crossed, the strain increases more rapidly for a small change in stress. In this region the material does not regain original dimension when the stress is removed. Thus, the material said to have a permanent set and the deformation is plastic deformation. This region is called as plastic region where Hooke's law is not valid.
4. If the material is unloaded beyond **b** say at **Point c**, the curve will proceed from Point **c** to **Point e**. If the material is loaded again from Point **e**, the curve will follow back to Point **c** with the same Elastic Modulus (slope) and thus shows perfect elastic behavior.
5. The material now has a higher yield strength of Point **c**. Increasing the yield strength of a material by permanently straining (deforming) it is called **Strain Hardening**. Strain Hardening takes place up to a maximum stress indicated by **f**. This largest value of stress which a material withstands without breaking is known **ultimate tensile strength**.
6. After **f**, material exhibits drastic increase in the strain for small or no increase in stress. The micro crack generates and propagates in the material due to continuous concentration of stress which results in fracture of material (**failure**) at **d**.

Hooke's law

Hooke's law states that when a material is loaded within its elastic limit, ***stress is directly proportional strain***.

It means that the ratio of stress to strain is constant within the elastic limit. This constant is known as Modulus of elasticity.

$$\frac{\text{stress}}{\text{strain}} = \text{constant or modulus of elasticity}$$

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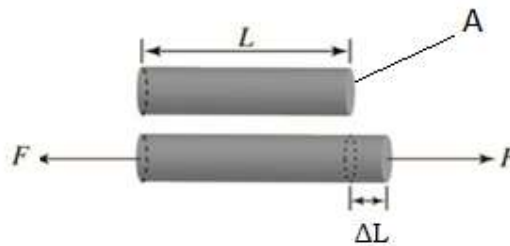
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$$\frac{\text{stress}}{\text{strain}} = \text{constant or modulus of elasticity}$$

Types of Elastic Moduli:

1. Young's Modulus of Elasticity (Y):

When a wire is acted upon by two equal and opposite forces in the direction of its length, the length of the body is changed. The change in length per unit length ($\Delta L/L$) is called the longitudinal strain and the restoring force (which is equal to the applied force in equilibrium) per unit area of cross-section of wire is called the longitudinal stress.



For small change in the length of the wire, the ratio of the longitudinal stress to the corresponding strain is called the **Young's modulus of elasticity (Y)** of the wire.

$$\text{Thus, } Y = \frac{\text{Longitudinal stress}}{\text{Linear Strain}} = \frac{(F/A)}{(\Delta L/L)} = \frac{FL}{A\Delta L}$$

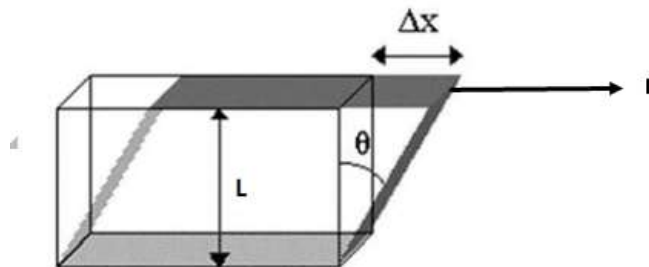
Let there be a wire of length 'L' and radius 'r'. It's one end is clamped to a rigid support and a mass M is attached at the other end. Then

$$F = Mg \text{ and } A = \pi r^2$$

Substituting in above equation, we have,

$$Y = \frac{MgL}{(\pi r^2)\Delta L}$$

2. Modulus of Rigidity (n or η):



When a body is acted upon by an external force tangential to a surface of the body, the opposite surfaces being kept fixed, it suffers a change in shape of the body, and its volume remains unchanged. Then the body is said to be sheared. The tangential force acting per unit area of the surface is called the '**shearing stress**' (F/A).

The ratio of displacement to perpendicular distance between the two surfaces is known as **shearing strain** (θ).

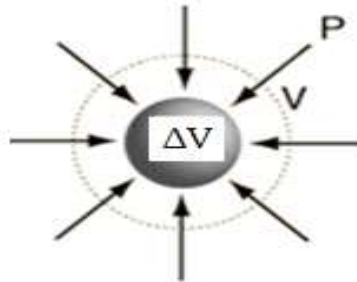
Shearing strain $\theta = \frac{\Delta x}{L}$ when θ is small.

For small strain, the ratio of the shearing stress to the shearing strain is called the '**modulus of rigidity**' of the material of the body. It is denoted by ' n or η '.

Rigidity modulus (n or η) = Tangential stress / shear Strain

$$n = \frac{F/A}{\Delta x/L} = \frac{F/A}{\theta} = \frac{F}{A\theta}$$

3. Bulk Modulus of Elasticity (K)



Describes volumetric elasticity or the tendency of an object to deform in all directions when uniformly loaded in all directions. It is defined as volumetric stress over volumetric strain, and is the inverse of compressibility.

$$K = \frac{\text{Volumetric stress}}{\text{Volume strain}} = \frac{FV}{A\Delta V}$$

When a uniform pressure (normal force) is applied all over the surface of a body, the volume of the body changes. The change in volume per unit volume of the body is called the '**volume strain**' and the normal force acting per unit area of the surface (**pressure**) is called the normal stress or **volume stress**.

For small strains, the ratio of the volume stress to the volume strain is called the '**Bulk modulus**' of the material of the body. It is denoted by K .

Then,
$$K = \frac{-P}{\frac{\Delta V}{V}}$$

Negative sign in formula implies that when the pressure increases volume decreases and vice-versa.

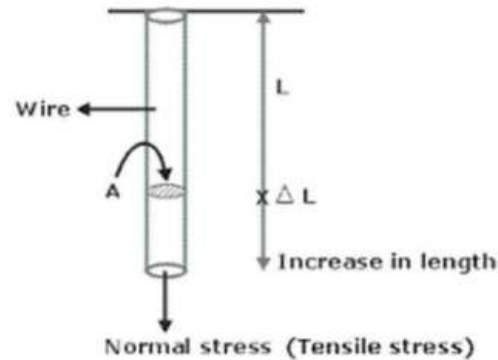
The reciprocal of the Bulk modulus of the material of a body is called the “**compressibility**” of that material. Thus,

$$\text{Compressibility} = 1/K$$

Longitudinal Strain Coefficient (α) :

“The longitudinal strain produced per unit stress is called longitudinal strain coefficient”.

$$\begin{aligned}\alpha &= \frac{\text{Longitudinal Strain}}{\text{Applied Stress}} \\ &= \frac{\frac{\Delta L}{L}}{T} = \frac{\Delta L}{TL}\end{aligned}$$



The extension produced due to the applied stress ‘T’ is $\Delta L = TL\alpha$

Lateral Deformation:

When there is a Longitudinal Strain in a material due to the deforming forces acting along the length, there is always a change in the thickness or diameter of the material. This change occurs in a direction perpendicular to the direction of the deforming force and is called lateral change.

Lateral stain:

If a deforming force acting on a wire of circular cross section with the original diameter ‘D’, produces a change ‘d’ in its diameter then,

$$\text{Lateral Strain} = \frac{d}{D}$$

Lateral Strain Coefficient (β):

The lateral strain produced per unit stress is called lateral strain coefficient.

$$\beta = \frac{\text{Lateral Strain}}{\text{Applied Stress}}$$

$$\beta = \frac{d}{D} = \frac{d}{TD}$$

Poisson's ratio (σ):

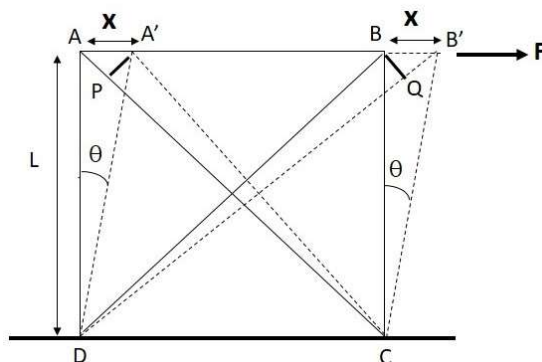
When a material is stretched, the increase in its length (Linear strain α) is accompanied by decrease in cross section (lateral strain β). Within the elastic limit, the lateral strain is proportional to longitudinal strain and the ratio between them is a constant for a material known as **Poisson ratio (σ)**.

$$\sigma = \frac{\text{lateral strain}}{\text{longitudinal strain}} \text{ i.e., } \sigma = \frac{\beta}{\alpha}$$

- **Relation between Shearing Strain, Longitudinal (Elongation) strain
compression strain:**

$$\text{Elongation strain} + \text{Compression strain} = \text{Shearing strain}$$

Expression for Young's Modulus (Y), Rigidity Modulus (n) and Poisson's Ratio (σ):



Consider a cube of side 'L', whose lower surface CD is fixed to a rigid support. Let a tangential force 'F' is applied at the upper surface along AB of the cube in a direction as shown

in figure. The applied tangential force causes the relative displacements at different parts of the cube, so that, A moves to A' and B moves to B' through a small angle 'θ'. Due to this, the diagonal AC will be shortened to A'C and diagonal DB will be increased to a length DB'. We have an extension Strain along DB and compression strain along AC. The angle 'θ' is the angle of shear (shearing strain) which is very small in magnitude.

The shearing stress acting on the body, $T = \frac{F}{A} = \frac{F}{L^2}$

From figure, shearing strain, $\theta = \frac{x}{L}$ ----- (1)

Shearing stress along AB is equivalent to elongation stress along BD and compressive stress along AC. Let α be the longitudinal strain coefficient and β be the lateral strain coefficient respectively. Then

Elongation along BD due to the tensile stress along BD = $BD \cdot T \cdot \alpha$

Elongation along BD due to the compressive stress along AC = $BD \cdot T \cdot \beta$

Total extension along BD is

$$QB' = BD \cdot T (\alpha + \beta) = \sqrt{2}LT (\alpha + \beta) \quad \text{---- (2) [since } BD = \sqrt{2}L]$$

Also, from right angled triangle QBB',

$$QB' = BB' \cos 45^\circ$$

$$QB' = \frac{x}{\sqrt{2}} \quad \text{----- (3)}$$

Eqn (2) becomes, $\frac{x}{\sqrt{2}} = \sqrt{2}LT (\alpha + \beta)$

$$\frac{x}{LT} = 2 (\alpha + \beta)$$

$$\frac{\theta}{T} = 2 (\alpha + \beta) \quad \text{(since from (1))}$$

Taking the reciprocal,

$$\frac{T}{\theta} = \frac{1}{2(\alpha + \beta)}$$

But $\frac{T}{\theta} = \eta$, the rigidity modulus

Therefore, $\eta = \frac{1}{2(\alpha + \beta)}$ ----- (4)

Young's modulus, $Y = \frac{\text{Longitudinal Stress}}{\text{Linear strain}}$

$$= \frac{1}{\frac{\text{Linear Strain}}{\text{Longitudinal Stress}}} = \frac{1}{\alpha}$$

Therefore, $Y = \frac{1}{\alpha}$ ----- (5)

This is the relation between young's modulus (Y) and linear strain (α).

We have from (4); $2(\alpha + \beta) = \frac{1}{\eta}$

$$2\alpha \left(1 + \frac{\beta}{\alpha}\right) = \frac{1}{\eta}$$

$$2\eta(1 + \sigma) = \frac{1}{\alpha} \quad \left(\because \frac{\beta}{\alpha} = \sigma\right)$$

$$\mathbf{2\eta(1 + \sigma) = Y} \quad \left(\because \frac{1}{\alpha} = Y\right)$$

The equation ($\mathbf{2\eta(1 + \sigma) = Y}$) gives relationship between Young's Modulus (Y), Rigidity Modulus (η) and Poisson's Ratio (σ).

Relationship between Bulk Modulus (K), Young's Modulus (Y), and Poisson's Ratio (σ):

$$K = \frac{Y}{3(1 - 2\sigma)}$$

$$\mathbf{Y = 3K (1 - 2\sigma)}$$

This is the **relation between Y, K and σ** .

Limits of Poisson's ratio (σ):

We have $Y = 3K(1 - 2\sigma)$ and $Y = 2\eta(1 + \sigma)$

Therefore, $3K(1 - 2\sigma) = 2\eta(1 + \sigma)$

- If σ be a positive quantity, $(1 - 2\sigma)$ should be positive
 $2\sigma < 1 \Rightarrow \sigma < 0.5,$

A perfect incompressible material deformed elastically at small strains would have a Poisson's ratio exactly 0.5.

- If σ be a negative quantity, $(1 + \sigma)$ should be positive. This implies that $\sigma > -1$

Thus, the limiting values of σ lies between -1 and 0.5. Negative value of σ would mean that on being extended, a body should also expand laterally. This hardly happens ordinarily. Similarly, a value of $\sigma = 0.5$ would mean that substance is perfectly incompressible. Generally, the limiting values of Poisson's ratio of different materials varies from -1 to 0.5

Factors Affecting Elasticity:

Some material will have change in their elastic property because of the following factors.

- a. Effect of stress
- b. Effect of annealing
- c. Change in temperature
- d. Presence of impurities
- e. Due to the nature of crystals

a) Effect of stress:

When a material is subjected to large number of cycles of stresses, it loses its elastic property even within the elastic limit. Therefore, the working stress on the material should be kept lower than the ultimate tensile strength and the safety factor.

b) Effect of Annealing:

Annealing is a process by which the material is heated to a very high temperature and then it is slowly cooled. Usually, this process is adopted for the material to increase the softness and ductility in the material. But annealing a material result in the formation of large crystal grains, which ultimately reduces the elastic property of the material.

c) Effect of Temperature:

The elastic property of the materials decreases with increase in the temperature due to decrease in the strength of inter molecular forces in materials with increase in temperature. But elasticity of invar steel (alloy) does not change with change of temperature.

Examples:

- The elastic property of lead increases when the temperature is decreased.
- The carbon filament becomes plastic at higher temperatures.

d) Effect of impurities:

The addition of impurities produces variation in the elastic property of the materials. The increase and decrease of elasticity depend upon the type of impurity added to it. If impurity added is more elastic than the material, then elasticity of the material increases and vice versa. Suitable impurities can alter the elastic properties of metals as they settle between the grains and brings connectivity between two grains.

Examples:

- When potassium is added to gold, the elastic property of gold increases.
- When carbon is added to molten iron, the elastic property of iron decreases provided the carbon content should be more than 1% in iron.
- If the carbon added is less 1% or in minute quantity, the elastic property of iron increases.

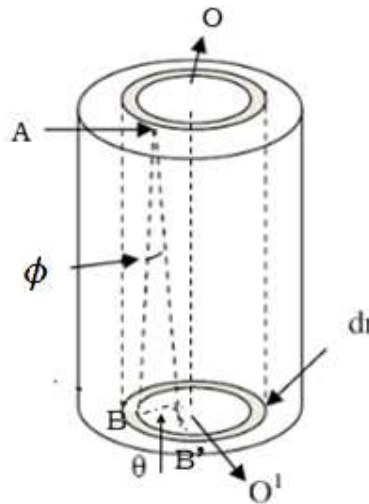
e) Nature of crystals

Elasticity of materials depends on crystalline nature of materials such as single crystals or poly crystals. Single crystals are more elastic than poly crystals due to the presence of grains in poly crystals.

Torsion of a Cylinder:

A long body which is twisted around its length as an axis is said to be under torsion. The twisting is brought into effect by fixing one end of the body to a rigid support and applying a suitable couple at the other end. The elasticity of a solid, long uniform cylindrical body under torsion can be studied, by imagining it to be consisting of concentric layers of the material of which it is made up of. The applied twisting couple is calculated in terms of the rigidity modulus of the body.

Expression for Couple per unit twist of a cylindrical rod



Consider a long cylindrical rod of length 'L' and radius 'R' rigidly fixed at its upper end.

$$\text{Rigidity modulus } \eta = \frac{\text{Shearing Stress}}{\text{Shearing Strain}}.$$

Therefore, Expression for Couple per unit twist of a cylindrical rod is given by,

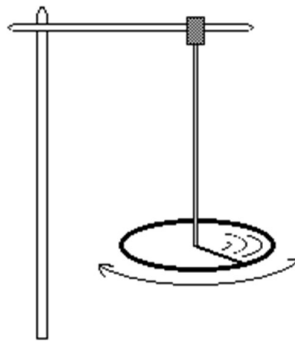
$$C = \left(\frac{\pi \eta R^4}{2L} \right)$$

Torsional Pendulum:

A heavy object suspended from end of a fine wire rotating about an axis constitutes a **torsional pendulum**.

Torsion pendulum consists of a heavy metal disc is suspended by means of a wire. When the disc is rotated in a horizontal plane so as to twist the wire, the various elements of the wire undergo shearing strain. The restoring couple of the wire tries to bring the wire back to the original position. As a result, the disc executes to and fro turning with the wire as the axis. These oscillations are known as torsional oscillations.

The oscillations executed by a suspended rigid body due to the twist in the suspension are known as **torsional oscillations**.



The time period of oscillation 'T' for a torsional pendulum is given by

$$T = 2\pi \sqrt{\frac{I}{C}}$$

Where I is the moment of inertia of the rigid body about the axis through the wire, C is the couple per unit twist for the wire.

Applications of Torsional Pendulum:

1. The moment of inertia of irregular rigid bodies can be determined using torsional pendulum.

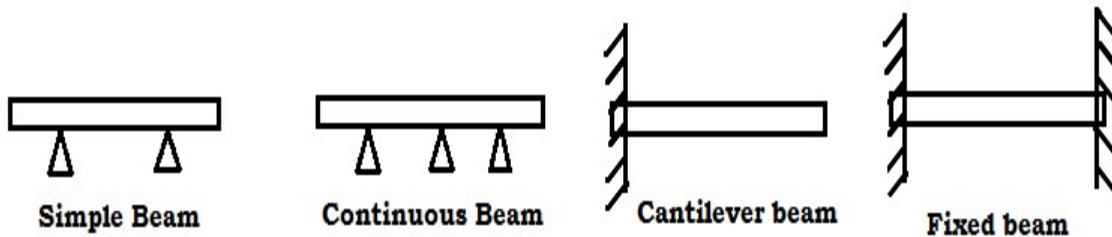
2. The rigidity modulus of a material can be found by taking the material in the form of wire and setting up of a Torsion pendulum.

Bending of Beam:

A homogenous body of uniform cross section whose length is large compared to its other dimensions is called a beam.

Types of Beams:

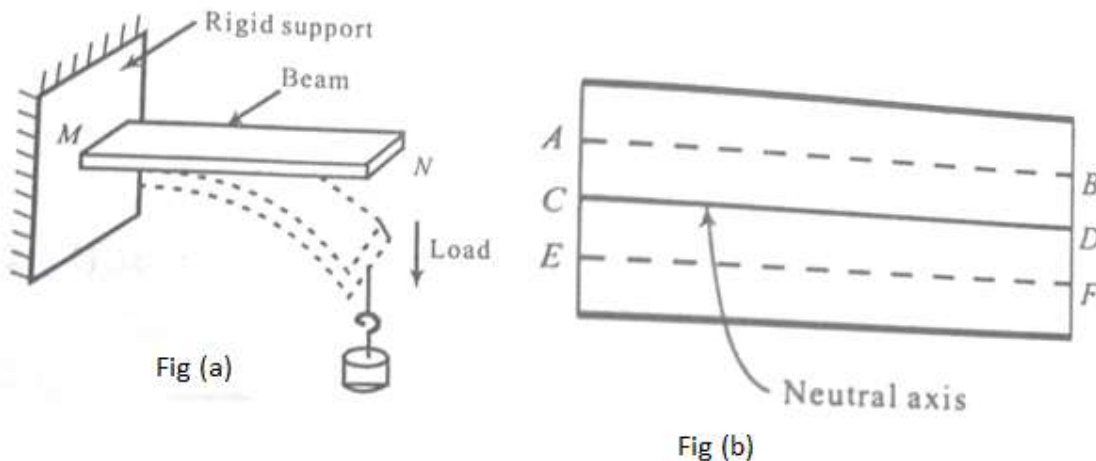
Depending on the support, beams are classified as following four types



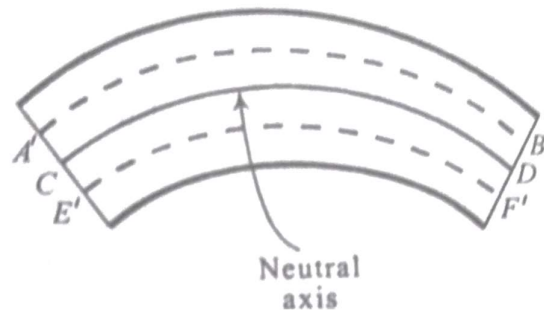
1. **Simple beam:** It is bar resting upon supports at its ends and is the most commonly used.
2. **Continuous beam:** It is a bar resting upon more than two supports.
3. **Cantilever beam:** It is a beam whose one end is fixed and the other end is free.
4. **Fixed beam:** A beam fixed at its both ends is called a fixed beam.

Neutral Surface and Neutral Axis:

Consider a uniform beam MN whose one end is fixed at M (Fig a). The beam can be thought of as made up of a number of parallel layers and each layer in turn as made up of a number of thin parallel longitudinal filaments or fibers in the plane of the layer. If a cross section of the beam along its length and perpendicular to these layers is taken the filaments of different layers appear like straight lines piled one above the other along the length of the beam (Fig b).



If a load is attached to the free end of the beam, the beam bends. The successive layers along with constituent filaments are strained. A filament like AB of an upper layer will be elongated to A^1B^1 and the one like EF of a lower layer will be compressed to E^1F^1 . The layer (CD) that does not undergo any change in the dimension is called as the **neutral layer** or **neutral surface**. A filament in the neutral layer whose length always remains the same is called the **neural axis**.



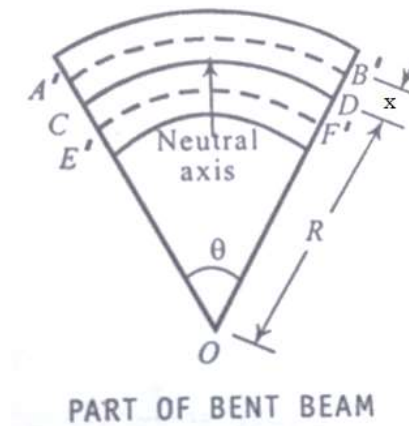
Neutral surface: is that layer of a uniform beam which does not undergo any change in its dimension when the beam is subjected to bending within the elastic limit.

Neutral Axis is an axis in the cross section of a beam along which there are no longitudinal stresses or strain. The length of this axis remains the same when the beam is subjected to bending.

Expression for bending moment of a beam:

Consider a uniform beam fixed at one end and loaded at the other. As a result, an equal reaction force acts in the upward direction at the fixed end. These two equal and opposite forces constitute a couple known as bending couple due to which the beam bends. The moment

(rotating effect) of the bending couple due to which a beam undergoes bending is called the **bending moment**.



Due to the bending, a layer like AB above the neutral layer will be elongated to A'B' and the one like EF below the neutral layer will be compressed to E'F'. CD is the neutral surface which does not undergo any change in its length. The layers of the bent beam form the part of concentric circles with centre at O. Let 'R' be the radius of the circle to which the neutral surface forms a part. Let ' θ ' be the angle subtended by the layers at the common centre 'O' and ' x ' be the separation between the successive layers.

Moment of the force acting on the entire layer

$$= \sum \frac{Yax^2}{R}$$

$$= \frac{Y}{R} \sum ax^2$$

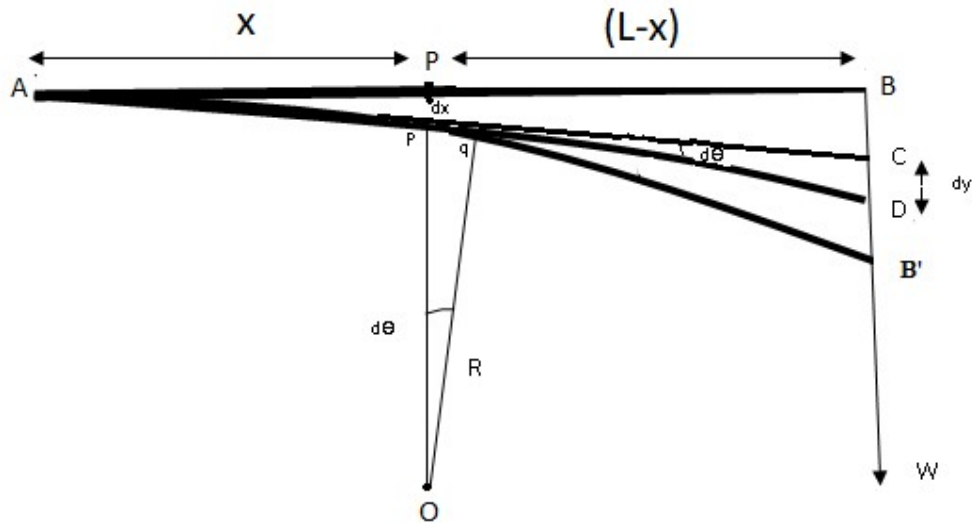
Here $\sum ax^2$ is called the **geometric moment of Inertia** I_g

i.e., $I_g = \sum ax^2 = AK^2$; where K is called the radius of gyration about the neutral axis.

Therefore, Moment of the force or bending moment $= \frac{Y}{R} I_g$

SINGLE CANTILEVER:

A beam fixed horizontally at one end and loaded at the other is called a **Cantilever**.



Consider a cantilever of length 'L' fixed at one end and loaded at the other end with a load 'W'. Let AB be the neutral axis of the cantilever. Consider a section P of the beam at a distance 'x' from A as shown in the figure.

$$\begin{aligned}\text{Bending moment about P} &= \text{Force} \times \text{perpendicular distance} \\ &= W(L - X) \quad \text{--- (1)}\end{aligned}$$

Due to this moment the beam bends such that the end B moves to B'.

$$\text{But bending moment of the beam} = \frac{Y}{R} I_g \quad \text{----- (2)}$$

Where R is the radius of curvature of neutral axis at P.

$$\text{From (1) and (2),} \quad W(L - X) = \frac{Y}{R} I_g \quad \text{----- (3)}$$

As the moment of the load increases towards the point A, the radius of curvature is different at different points and decreases towards A. For a point Q at a very small distance dx from P. Q is practically same as at P. let $d\theta$ be the angle subtended by P and Q at the centre 'O'

Therefore, $PQ = dx = R d\theta$

$$\Rightarrow R = \frac{dx}{d\theta} \quad \text{----- (4)}$$

$$\text{From (3), } W(L - X) = (Y I_g) \frac{d\theta}{dx}$$

$$\text{Or } d\theta = \frac{W(L-x)dx}{Y I_g} \quad \text{----- (5)}$$

Draw tangents to the neutral axis at P and Q meeting the vertical line BB' at C and D. The angle subtended by them is $d\theta$. The depression dy of Q below P is given by

$$CD = dy = (L - X)d\theta$$

Or $d\theta = \frac{dy}{L-x}$ ----- (6)

Substituting Eqn. (6) in Eqn.(5), $dy = \frac{w(L-x)^2}{YI_g} dx$ ----- (7)

Total depression BB¹ of the loaded end, $y = \int_0^L \frac{w(L-x)^2}{YI_g} dx$

$$= \frac{w}{YI_g} \int_0^L (L^2 + x^2 - 2Lx) dx$$

$$y = \frac{w}{YI_g} \times \frac{L^3}{3}$$

For rectangular cross section $I_g = \frac{bd^3}{12}$; b = breadth, d = thickness of beam

Then, the depression, $y = \frac{4mgL^3}{Ybd^3}$ [$\because W = mg$]

And Young's Modulus $Y = \frac{4mgl^3}{ybd^3}$

Applications of beam: Beams are used

1. In the fabrication of trolley ways.
2. In the Chassis/ frame as truck beds.
3. In the elevators.
4. In the construction of platform and bridges.
5. Beams are an integral part of Civil engineering structural elements (bridges, dams, multi-storeyed buildings).
6. As girders in buildings and bridges.

NUMERICALS:

- 1. Calculate the minimum diameter of the brass rod, if it is required to support a load of 600N without exceeding the elastic limit 3.8×10^8 Pa.**

Solution: Load or Force $F=600N$, stress at elastic limit $=T=3.8 \times 10^8 \text{ Pa} = 3.8 \times 10^8 \text{ N/m}^2$
 $(\because 1 \text{ Pa} = 1 \text{ N/m}^2)$

To find diameter of brass rod (d)

$$\text{w.k.t. stress } (T) = \frac{\text{Load}(F)}{\text{area}(a)} \text{ where } a = \frac{\pi}{4} d^2$$

$$\therefore a = \frac{F}{T}$$

$$\frac{\pi}{4} d^2 = \frac{600}{3.8 \times 10^8} \text{ or } d^2 = 2.01 \times 10^{-6}$$

$$\therefore d = \sqrt{2.01 \times 10^{-6}} = 1.417 \times 10^{-3} \text{ m}$$

- 2. Determine the elongation produced in a wire of length 2m and radius $0.013 \times 10^{-3} \text{ m}$, when subjected to an axial force of 14.7N. Take $Y = 2 \times 10^{11} \text{ N/m}^2$.**

Solution: Wire length = $L = 2 \text{ m}$, wire radius = $r = 0.013 \times 10^{-2} \text{ m}$

$$\therefore \text{ wire diameter } d = 2r = 2(0.013 \times 10^{-2}) = 0.026 \times 10^{-2} \text{ m}$$

$$\text{And area } a = \frac{\pi}{4} d^2 = \frac{\pi}{4} (0.026 \times 10^{-2})^2 = 5.31 \times 10^{-8} \text{ m}^2$$

$$\text{Force or load} = F = 14.7 \text{ N and Young's modulus} = Y = 2 \times 10^{11} \text{ N/m}^2$$

To find the elongation of wire (ΔL)

$$\text{w.k.t. } \Delta L = \frac{FL}{aY} = \frac{14.7 \times 2}{(5.31 \times 10^{-8})(2 \times 10^{11})} = 2.76 \times 10^{-3} \text{ m}$$

- 3. A rod of cross-sectional area $15 \text{ mm} \times 15 \text{ mm}$ and 1 m long is subjected to compressive load of 22.5 kN . Calculate the stress and decrease in length if Young's modulus is 200 GN/m^2 .**

$$\text{Solution: cross-sectional area} = a = 15 \times 15 \text{ mm} = 225 \text{ mm}^2 = 225 \times 10^{-6} \text{ m}^2$$

$$\text{Length } L = 1 \text{ m; load } F = 22.5 \text{ kN} = 22.5 \times 10^3 \text{ N}$$

$$\text{Young's modulus of elasticity } Y = 200 \text{ GN/m}^2 = 200 \times 10^9 \text{ N/m}^2 \quad \text{Giga(G)} = 10^9$$

To find stress (T)

$$\text{w.k.t. stress } T = \frac{\text{load}(F)}{\text{area}(a)} = \frac{22.5 \times 10^3}{225 \times 10^{-6}} = 1 \times 10^8 \text{ N/m}^2$$

To find decrease in length (ΔL) of rod

$$\text{w.k.t. strain } \epsilon = \frac{\Delta L}{L} \quad \therefore \Delta L = \epsilon L \quad \dots\dots\dots (1)$$

But strain $\epsilon = ?$

$$\text{w.k.t. Young's modulus } Y = \frac{T}{\epsilon} \quad \text{or} \quad \epsilon = \frac{T}{Y} = \frac{1 \times 10^8}{200 \times 10^9} = 5 \times 10^{-4}$$

Now, equation (1) becomes $\Delta L = (5 \times 10^{-4})(1) = 5 \times 10^{-4} \text{ m}$ or 0.5 mm

Note: Decrease in length can also be calculated using $\Delta L = \frac{FL}{aY}$

- 4. Calculate the force required to produce an extension of 2 mm in a steel wire of length 2m and diameter 1mm. The Young's modulus the material is $2 \times 10^{11} \text{ N/m}^2$.**

Solution: extension $= \Delta L = 1 \text{ mm} = 1 \times 10^{-3} \text{ m}$ wire length $= L = 2 \text{ m}$

Wire diameter $= d = 1 \text{ mm} = 1 \times 10^{-3} \text{ m}$

$$\therefore \text{c/s}^{\text{al}} \text{ area of wire} = a = \frac{\pi}{4} d^2 = \frac{\pi}{4} (1 \times 10^{-3})^2 = 7.85 \times 10^{-7} \text{ m}^2$$

Young's modulus of steel wire $= Y = 2 \times 10^{11} \text{ N/m}^2$

To find force (F) to produce extension (ΔL)

$$\text{w.k.t. extension} = \Delta L = \frac{FL}{aY}$$

$$\therefore \text{Force } F = \frac{(\Delta L)aY}{L} = \frac{(1 \times 10^{-3})(7.85 \times 10^{-7})(2 \times 10^{11})}{2} = 78.5 \text{ N}$$

- 5. A rubber tube 0.4m with external and internal diameters 0.01m and 0.04 m respectively extends by 0.0006m under an axial force of 5kg weight. Calculate the Young's modulus of the rubber material.**

Solution: The given rubber tube is a hollow material with external and internal diameters.

Note that in all the previous problems, the component material was solid in nature.

Rubber tube length = $L = 0.4m$, tube external diameter = $d_o = 0.01m$

Tube internal diameter = $d_i = 0.004m$

$$\therefore \text{Cross-sectional area of tube} = a = \frac{\pi}{4}(d_o^2 - d_i^2) = \frac{\pi}{4}(0.01^2 - 0.004^2) = 6.59 \times 10^{-5} m^2$$

$$\text{Extension} = \Delta L = 0.0006m \quad \text{Force} = F = 5Kg = 5 \times 9.81N = 49.05N$$

To find Young's modulus (Y)

$$\text{w.k.t. extension} = \Delta L = \frac{FL}{aY}$$

$$\therefore Y = \frac{FY}{\Delta L \times a} = \frac{49.05 \times 0.4}{(0.0006)(6.59 \times 10^{-5})} = 496.2 \times \frac{10^6 N}{m^2}$$

6. Calculate the Poisson's ratio for Silver, having Young's modulus as $7.25 \times 10^{10} N/m^2$ and bulk modulus $11 \times 10^{10} N/m^2$.

Solution: Young's modulus = $Y = 7.25 \times 10^{10} N/m^2$

Bulk modulus = $K = 11 \times 10^{10} N/m^2$

To find Poisson's ratio (σ)

$$\text{w.k.t. } Y = 3K(1 - 2\sigma)$$

$$7.25 \times 10^{10} = 3(11 \times 10^{10})(1 - 2\sigma)$$

$$(1 - 2\sigma) = 0.219 \quad \text{or} \quad 2\sigma = 0.78 \quad \text{or} \quad \sigma = 0.39.$$

7. The Young's modulus for steel is $2 \times 10^{11} N/m^2$ and Rigidity modulus $8 \times 10^{10} N/m^2$. Calculate the Poisson's ratio and bulk modulus of steel.

Solution: Young's modulus = $Y = 2 \times 10^{11} N/m^2$ Rigidity modulus = $n = 8 \times 10^{10} N/m^2$

To find Poisson's ratio (σ)

$$\text{w.k.t. } Y = 2n(1 + \sigma)$$

$$2 \times 10^{11} = 2(8 \times 10^{10}) (1 + \sigma) \quad \text{or} \quad (1 + \sigma) = 1.25 \quad \text{or} \quad \sigma = 0.25$$

To find Bulk modulus(K)

$$\text{w.k.t. } Y = 3K(1 - 2\sigma)$$

$$2 \times 10^{11} = 3K [1 - 2(0.25)] \quad \text{or} \quad 2 \times 10^{11} = 3K(0.5)$$

$$\therefore K = \frac{2 \times 10^{11}}{3(0.5)} = 1.33 \times 10^{11} \text{ N/m}^2$$

8. The Young's modulus for a material is $100 \times 10^9 \text{ N/m}^2$ and its modulus of rigidity is $40 \times 10^9 \text{ N/m}^2$. Determine its bulk modulus for the given material.

Solution: Young's modulus = $Y = 100 \times 10^9 \text{ N/m}^2$ Rigidity modulus = $n = 40 \times 10^9 \text{ N/m}^2$

$$\text{w.k.t. } Y = 3K(1 - 2\sigma) \quad \dots\dots\dots(1)$$

But Poisson's ratio $\sigma = ?$

$$\text{w.k.t. } Y = 2n(1 + \sigma)$$

$$100 \times 10^9 = 2(40 \times 10^9)(1 + \sigma) \quad \text{or} \quad 1 + \sigma = 1.25 \quad \text{or} \quad \sigma = 0.25$$

Now equation (1) becomes $100 \times 10^9 = 3K [1 - 2(0.25)]$

$$\text{Or } K = \frac{100 \times 10^9}{3(0.5)} = 66.67 \times 10^9 \text{ N/m}^2$$

Note: Bulk modulus (K) can also be calculated using the relation $Y = \frac{9nK}{3K + n}$

9. A solid sphere of radius 10.3m is subjected to a normal pressure of 10 N/m^2 all over its surface. Determine the change in volume of the sphere if the bulk modulus of the material is $4.58 \times 10^{10} \text{ N/m}^2$.

Solution: Sphere radius = $r = 10.3 \text{ m}$ applied pressure = $P = 10 \text{ N/m}^2$,

Bulk modulus = $K = 4.58 \times 10^{10} \text{ N/m}^2$

$$\text{w.k.t. Bulk modulus} = K = \frac{\text{identical axial stress } (T)}{\text{Volumetric strain } (\frac{\Delta B}{V})} \quad \dots\dots (1)$$

where V = volume of sphere = $\frac{4}{3}\pi r^3 = \frac{4}{3}\pi(10.3)^3 = 4.577 \times 10^3 \text{ m}^3$

Equation (1) can be written as, $K = \frac{P}{\left(\frac{\Delta V}{V}\right)}$ or $K = \frac{PV}{\Delta V}$ or $\Delta V = \frac{PV}{K} = \frac{10(4.577 \times 10^3)}{4.58 \times 10^{10}} = 10 \times 10^{-7}$

or change in volume of sphere = $\Delta V = 1 \times 10^{-6} \text{ m}^3$